

# **Riverside Research Plasma Facility**

### TAT A GLANCE



Artificial Intelligence & Machine Learning

CORE RESEARCH AREAS

BROAD SPECTRUM OF SUBJECT MATTER EXPERTS

Electromagnetics

**Plasma Physics** 

Radar

Software Engineering

System Engineering

**Tursted & Resilient Systems** 

THz Imaging

41,000 ACCREDITED OFFICES, LABS, AND NETWORKS Lexington, MA Remote Location

FOUNDED IN 1967 | CHARTERED TO ADVANCE SCIENTIFIC RESEARCH FOR THE BENEFIT OF THE NATION



- configuration.
- DC OPERATION

Magnet	Curren
1	99.8 A
2	99.5 A
3	99.1 A
4	99.8 A
5	99.9 A







25 m<sup>3</sup> Pumping Capacity: 11,800 L/s • 4 turbomolecular pumps • 2 cryopumps

Large Chamber

- Create a large, flexible environment for various
- experimental campaigns Significant control over plasma density, background magnetic field, plasma species
- Upgrading diagnostic suite to improve plasma characterizatior capabilities



- > 0.4 m<sup>3</sup>
- Significant optical and diagnostic access available
- Base pressure ~1uTorr
- Test bench currently used for plasma-material interaction studies
- See poster GP12.00015 for information on ongoing experiments.



### **Small Chamber**

- > 0.03 m<sup>3</sup>
- Highly customizable, each panel replaceable with: windows
- feedthrough ports
- single ISO250 port
- optical breadboards
- Attachable to larger chambers to increase pumping capacity while maintaining quick vent sessions

### **Experimental Equipment**

### **CURRENT:**

- > LaB6 thermionic plasma source
- Microwave Plasma Sources
- 2.54 GHz at 1000 W • 0.7 – 2.7 GHz at 100 W
- 6 18 GHz at 10 W
- LPKF Protomat mill for custom PCBs
- > 2 J/pulse Nd:YAG laser (1064/532 nm)
- > 3 650 GHz coherent radiation sources (VDI)
- > 3 stage additive/subtractive spectrometer
- > 1.7 Farad capacitor bank
- > 6 45 KW power source banks
- > 120 kW power supply
- > Langmuir Triple probe



- > RFA probes
- > Residual gas analyzers
- > Supercontinuum MIR laser > Vacuum compatible motorized stages
- > Hall thruster
- **PLANNED:**
- > LIF system
- > Mach probes
- Advanced interferometry system
- > RF plasma source

**TECHNICAL EXCELLENCE** AND TRUSTED SOLUTIONS FOR THE DEFENSE AND INTELLIGENCE COMMUNITIES

## **Benchmarking LaB<sub>6</sub> Operation in the Large Chamber**



- LaB<sub>6</sub> thermionic emitter operated in large chamber. Four 4"x4" LaB<sub>6</sub> tiles with rotationally symmetric
- tongue-in-groove joints. 1"x8" slab graphite mask easily restricts plasma
- source area. • More masks (8" circular, 7.875" square)
- available to vary plasma source area. Operated plasma source in a variety of magnetic field configurations, gas species, and cathode temperatures.

Graphite heating element can source up to 800 A Can heat LaB<sub>6</sub> up to  $\sim 2000$  C.

Measure cathode temperature with a 2-color pyrometer. **PULSED OPERATION** 

In pulsed mode, trigger sent to 1.7 F cap bank, applying differential voltage between cathode and anode. Differential voltage applied for 20 ms, 1 Hz repetition

Cap bank can supply up to 5 kA of current in current

Reproducible plasma discharges observed in preliminary operation.

In DC mode, trigger pulls voltage from power supply to cathode and anode. Supplies voltage between 0-330 V and current from 0-330 A. Plasmas last between several seconds and several minutes. Useful testbed for turbulence studies.

## **Uniform Field Operation**







- $\rightarrow$  96 turns each.
- 45 kW available for each magnet. Can operate with on-axis field up to 0.3 T.
- Individually controllable to customize magnetic field profile.
- Mass flow controller can operate up to 5000 SCCM.
- Can operate with any nonreactive species.
- MFC and pumping capacity can be tailored for specific background pressures.

#### LARGE CHAMBER PRIMARY EXPERIMENTAL GOALS, **TO DATE:**

#### **Ionospheric research**

Match nondimensional plasma characteristics of ionosphere. Explore plasma instabilities that can affect satellite communications.

- Hypersonic sheath characterization Creates dense, low field plasma and test standard communications
- Explores mitigation schemes.
- **Plasma-based thrusters**
- Electrothermal and electromagnetic thrusters.

FUTURE EXPERIMENTAL GOAL: Plasma-material interaction studies.



## Low Field Operation.

	Magnet         1         2         3         4         5	Current 99.8 A 9.63 A 8.96 A 9.86 A 9.84 A	Voltage         7.8 V         0.76 V         0.71 V         0.77 V         0.78 V	B-Field Component Maximum (Solver) Abs Maximum (Solver) Abs Abs Abs Abs Abs Abs Abs Abs Abs Abs	Measurement re	egion	$ \begin{array}{c} 150 \\ 140 \\ 130 \\ 120 \\ 120 \\ 110 \\ 100 \\ 90 \\ 90 \\ 80 \\ 70 \\ 40 \\ 50 \\ 40 \\ 20 \\ 10 \\ 0 \\ \end{array} $
Helium, 1800 C, D	C Dischar	<u>rge</u> ×10 4	17	<b>Helium, 1800 C</b>	×10 <sup>17</sup>	0	Helium, 1800 C
		3.5	n <sub>e</sub> (m <sup>-3</sup> ) Docition (mm)	50 100 150 200	4 3 °,() 2 ⊂ <sup>⊕</sup> 1	(mm) 100 150	
0 50 100 Height (n	150 וm)	200		0 10 20 Time (ms)	30	0	10 20 Time (ms)
/lirror F	ie	<b>d</b> (	Op	eration	G 600 ★		
MagnetCurrent1500 A	Voltage 39.8 V				550 — 500 — 450 —		



### **Riverside Research Plasma Facility Takeaways**

### Large Chamber LaB6 plasma benchmarking

- LaB<sub>6</sub> thermionic emitter run in a variety of regimes.
- Cathode mask used to match a desired plasma structure.
- Future runs will be done with either a circular mask (8" diameter) or a large square mask (7.875" edge length) to form larger diameter plasmas. Can use arbitrary shaped masks to explore other physics effects.
- Individually controllable magnets significantly impact both plasma shape and dynamics.
- Uniform field led to a relatively narrow slab plasma.
- Low field led to a more diffuse, less dense plasma.
- Mirror field led to a dense plasma with significant instability activity. Helium and Argon discharges explored in a variety of run conditions.
- Cathode temperature can be continuously varied up to ~2000 C.
- Plasma was consistently run at 1400 C.
- Plasma was significantly denser at 1800 C.
- Plasmas can be run repeatably, consistently, with a relatively short turnaround.
- The large plasma chamber can accommodate a wide range of experimental regimes. Plan on expanding diagnostics available for general operation.
- Currently use LTP on a motorized stage, mid speed camera.
- Expanding to Bdot probes, Mach probes, RFA, LIF, spectroscopy, interferometry.
- **Riverside Research facilities are open to contracts with outside partners.** Please contact Dr. Stephen Parsons at sparsons@riversideresearch.org for additional information or work opportunities.

### Neutralizer study

- Small chamber is currently attached to the large chamber behind a gate valve.
- Gives access to the pumping capacity of the large chamber while allowing for quick up-to-air periods to change experimental setup.
- Can vent small chamber, change a setup, and be back to UHV in ~15 minutes.
- Running split ring resonator as plasma source for an electrothermal thruster neutralizer • Achieving electron density on order 3\*10<sup>16</sup> m<sup>-3</sup> for low pressure operation.
- Higher density operation available at higher pressures (~atmosphere)
- Current efficiency ~20 mA/W.
- Microstrip resonator attached to a flowing gas line has potential applications as a general plasma source.

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